Remeshing-Free Graph-based Finite Element Method for Fracture Simulation

A. Mandal¹ and P. Chaudhuri² and S. Chaudhuri¹

¹ Department of Electrical Engineering, Indian Institute of Technology Bombay, India {avirupmandal, sc}@ee.iitb.ac.in
² Department of Computer Science and Engineering, Indian Institute of Technology Bombay, India {paragc}@cse.iitb.ac.in



Figure 1: Our novel fracture simulation algorithm produces the intricate fracture patterns that result from the tearing of a loaf of bread.

Abstract

Fracture produces new mesh fragments that introduce additional degrees of freedom in the system dynamics. Existing finite element method (FEM) based solutions suffer from increasing computational cost as the system matrix size increases. We solve this problem by presenting a graph-based FEM model for fracture simulation that is remeshing-free and easily scales to high-resolution meshes. Our algorithm models fracture on the graph induced in a volumetric mesh with tetrahedral elements. We relabel the edges of the graph using a computed damage variable to initialize and propagate fracture. We prove that non-linear, hyper-elastic strain energy density is expressible entirely in terms of the edge lengths of the induced graph. This allows us to reformulate the system dynamics for the relabeled graph without changing the size of the system dynamics matrix and thus prevents the computational cost from blowing up. The fractured surface has to be reconstructed explicitly only for visualization purposes. We simulate standard laboratory experiments from structural mechanics and compare the results with corresponding real-world experiments. We fracture objects made of a variety of brittle and ductile materials, and show that our technique offers stability and speed that is unmatched in current literature.

 $\begin{array}{l} \text{CCS Concepts} \\ \bullet \textit{Computing methodologies} \rightarrow \textit{Physical simulation;} \end{array}$

1. Introduction

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Fracture is a ubiquitous phenomenon in the real world. Everyday we come across several instances of fracture, ranging from the shattering of a brittle glass tumbler to the tearing of a soft loaf of bread. To recreate such real-world phenomena, realistic dynamic fracture simulations are used frequently in computer graphics. Simulating intricate fracture patterns and crack propagation often requires complex mathematical formulation. Incorporating both brittle and ductile fracture in a single framework needs even more sophisticated analysis. It has thus been an important research topic in both structural mechanics and computer graphics.

However, despite intensive research over the last two decades, the existing fracture techniques still suffer from scalability issues, poor stability and high computational cost. E.g., as the number of cracks grows in an object, the Finite Element Method (FEM) based approaches proposed in [OBH02] and [OH99] require remeshing to address the additional degrees of freedom (DOF). This poses challenges like the generation of thin (or sliver) elements, instability and heavy computational cost. To avoid these effects authors have explored remeshing-free techniques like eXtended Finite Element Method (XFEM) [KBT17] [CMSK20] and the Material Point Method (MPM) [PKA*05] [WDG*19] [HFG*18] [WFL*19] [WCL*20]. The system matrix of XFEM scales linearly with the introduction of new cracks, which in turn requires more computation time. Moreover, in XFEM large volume ratios in Heaviside enrichment leads to severe stability issues. On the other hand, the particlebased material point methods have shortcomings with respect to imposing boundary conditions, high computational cost and rigidity of fragmented segments. Contrary to the discrete formulations, peridynamics-based methods [CZZ*18] [LBC*15] and boundary element methods (BEM) [HW15] [HW16] deal with solving the integral equations of continuum mechanics, thus bypassing the difficulty of imposing boundary conditions. But solving these singular integrals in peridymanics is extremely expensive in computation and therefore, is not suitable for low-resource applications. The use of BEM is limited to materials with large volumes simulating only brittle fracture.

We perform our simulations on volumetric meshes with tetrahedral elements. This underlying mesh, on which all computation is performed, is called the computational mesh. This mesh induces a graph where its vertices become the nodes of the graph and the edges of the mesh become the edges of the graph. The computational mesh is never remeshed. This implies that, unlike XFEM, the number of degrees of freedom of the system does not change with the introduction of cracks, i.e., the size of the initial system matrix does not change. This allows our method to scale to high resolution models with very little extra computational overhead. The mesh that is rendered for visualization is the same as the computational mesh initially. This mesh, however, needs to be split and the fracture surfaces have to be reconstructed for rendering the fracture. This does not affect the computational mesh.

It has been shown in earlier work that the nodal forces of a discretized hyper-elastic FEM system can be completely represented in terms of a function of strain energy density along the edges of the induced graph [RS15]. We use this fact and follow prior work in structural mechanics [KRS16] to define a purely edge-based damage variable, which describes the extent of the damage for any edge in the graph. We build over prior work to reformulate the strain energy density of damaged elements in 3D volumetric object meshes and then simulate the independent movement of the fractured pieces. Further in our model, we can generate and control the diffusion of cracks into the object by imposing a non-local fracture criterion.

We demonstrate the efficacy and robustness of our method by simulating realistic examples of both brittle and ductile fracture. Our method's ability to handle fracture of a variety of materials ranging from stiff glass to squishy jello makes it useful for a wide range of applications.

Our contributions are summarized as follows:

- We propose a novel remeshing-free, graph-based, computationally efficient and robust FEM solution for fracture simulation of 3D models.
- We present a generalized version of a graph-based FEM algorithm that can simulate fracture of materials with both linear and non-linear strain energy density.
- We theoretically prove that hyper-elastic strain energy density can be represented in closed form as a function of the length of the edges of any object mesh.
- We extensively validate the correctness of our model against real-world structural mechanics experiments.
- We demonstrate via examples that our method serves as a unified FEM framework for simulating both brittle and ductile fracture.

The rest of the paper is organized as follows. We start by presenting a discussion of related works in Section 2. Subsequently, in Section 3 we first describe the theoretical formulation of graphbased FEM, and then detail the technical aspects of fracture via crack initiation and crack propagation. We discuss the implementation details about the visualization of fractured material, collision handling and the complete algorithm for our model in Section 4. Next, we present results for various kinds of materials undergoing fracture, generated using our method in Section 5. To validate our method and check its correctness, we compare our results with realworld benchmark fracture experiments and existing fracture simulation techniques in the literature. Finally, we conclude our paper by discussing the limitations of our work and future directions.

2. Related Work

We first delve into existing methods for mesh-based fracture simulation in computer graphics. Next, we explore meshless particlebased fracture simulation methods. We conclude the section by reviewing the graph-based finite element analysis presented in material science literature.

2.1. Mesh-based Fracture Simulation

The inception of fracture simulation in computer graphics goes back to the seminal work on visco-elastic fracture by Terzopoulos and Fleischer [TF88]. Early approaches propose to model brittle fracture by removing the springs in a mass-spring system [ADKK04] [NTB*91] [HTK00] [HTK98] depending on a stressbased yield threshold. But in such a system, where point masses are connected by springs, the sudden removal of springs leads to significant visual artefacts. Moreover, visualization of the crack surface often requires the use of a tetrahedral marching algorithm and is therefore computationally expensive [HTK00] [HTK98]. Later, finite element method based solutions for simulating fracture have



(a) The model just prior to fracture. (b) Strain profile just prior to fracture.

(c) Model post fracture.

(d) Strain profile post fracture.

Figure 2: Here we show frames from a simulation where we rip off the limbs of a jello armadillo. In the strain profiles, the red to blue colour gradient denotes the highest to lowest strain values. The model fractures where the strain is highest prior to fracture. Post fracture, the strain drops.

been more successful and different variations of those approaches are still widely used. The first breakthrough in FEM-based fracture came in the paper by O'Brien et al. [OH99]. In their work, the authors present a nodal stress-based analysis for brittle fracture that was later extended further to ductile fracture [OBH02] [MG04]. Bao et al. [BHTF07] also present a method to simulate both brittle and ductile fracture. These FEM-based fracture techniques rely on splitting the elements of a tetrahedral mesh, which satisfy the fracture conditions and then remesh the entire fractured mesh with the crack opening. However, these approaches suffer from various geometric difficulties like remeshing near crack tips, generation of degenerate elements and forming ill-conditioned basis matrices. Subsequent FEM-based approaches employ different techniques to alleviate these problems, such as dynamic local mesh refinement to repair degenerate tetrahedra [WRK^{*}10], remeshing depending on gradient descent flow for finer fracture resolution [CYFW14], adaptive subdivision schemes for tetrahedral [KLB15] and triangular meshes [PNdJO14] where remeshing is done around the high stress areas to improve fracture resolution.

FEM-based solutions that work with a virtual node algorithm (VNA) [MBF04] duplicate elements and add extra degrees of freedom to facilitate partial or full crack openings, instead of splitting the mesh elements. Later, VNA-based methods are improved to incorporate cutting at lower than mesh resolution [SDF07] and to robustly handle intersections passing through a node, edge or face through an adaptive element duplication approach [WJST15]. Recently, Koschier et al. [KBT17] presented a remeshing-free extended finite element method (XFEM) based algorithm to simulate dynamic pre-defined cuts in a 3D mesh. Like VNA, XFEM too adds extra degrees of freedom using enrichment functions but the advantage of XFEM over VNA is while VNA duplicates the mesh element leading to erratic conservation of mass, XFEM splits the initial mass of an element into fragments. Chitalu et al. [CMSK20] expanded the pre-defined cuts to crack generation and propagation.

Contrary to processing fracture in volume elements like FEM, boundary element methods (BEM) simulate cracks on a surface mesh. Even though BEM found its way into graphics for simulating elastostatic deformable objects James et al. [JP99], it has been used recently for fracture simulation. Hahn et al. [HW15] and Zhu et al. [ZBG15] used BEM with stress intensity factors (SIFs) along crack fronts to successfully simulate brittle fracture. Later Hahn et al. [HW16] expanded their work to develop an algorithm for fast approximation of BEM brittle fracture, alleviating the high computational cost for solving singular integrals. Currently, to the best of our knowledge, there exists no work on BEM-based ductile fracture in graphics.

Peridynamics methods solve the integral equations of continuum mechanics. Unlike discrete differential-based methods e.g., FEM or MPM, which need to impose necessary boundary conditions for simulating fracture, it is trivial to model discontinuities using integral-based peridynamics. In initial work using these ideas, authors [LBC*15] approximate the behaviour of peridynamics using mass-spring systems to simulate brittle fracture. Later work [CZZ*18] presents original peridynamics-based fracture algorithms to simulate brittle as well as ductile fracture.

2.2. Meshless Fracture Simulation

Among the meshless methods, material point methods have gained considerable recognition in recent years. The first major breakthrough of MPM in computer graphics came through the seminal work in snow simulation by Stomakhin et al. [SSC*13]. Later MPM found its application in simulating a wide range of materials as such chocolate, lava, hybrid fluids and rubber to name a few. Interested readers can look at [JST*16] for a detailed overview of different MPM methods. Most recently, MPM has shown promising results in simulating fracture due to its ability to handle extreme topological change. Wu et al. [HFG*18] propose the Compatible Particle-in-Cell (CPIC) algorithm, which allows the simulation of sharp discontinuities inside a material and thus simulates dynamic material cutting. Later Wolper et al. [WFL*19], introduce Continuum Damage Mechanics (CDM) with a variational energy-based formulation for crack evolution. This achieves a realistic dynamic fracture simulation for isotropic materials. Further study [WCL*20] devoted to formulating non-local CDM to simulate anisotropic material fracture.

MPM methods reproduce realistic fracture phenomena but they are not well suited to rendering sharp boundaries. They not only have difficulty in enforcing essential boundary conditions, but they also suffer from the disappearance of intricate crack patterns due to excessive smoothing, high computational cost and rigidity of fragmented elements. Further, MPM methods struggle to simulate rigid object fracture.

2.3. Graph-based Finite Element Method

While simulating fractures in the material domain, classical FEM introduces degenerate elements due to remeshing and XFEM has several limitations like ambiguities of crack-tip enrichment in 3D, instabilities induced by large volume ratios in Heaviside enrichments, difficulties of handling branch enrichments in 3D. Moreover, in both of these methods, computational cost increases linearly with an increase in the number of cracks. While BEM and peridynamics overcome these problems by solving direct integrals, the use of BEM is limited to materials with large volumes simulating only brittle fracture and peridynamics has a high computational cost. Meshless methods have their limitations as described in the previous section.

Work in finite element analysis literature by Reddy and Srinivasa [RS15] proposes a solution based on classical FEM to show that for any hyper-elasticity problem, the magnitude of the nodal forces can be written completely in terms of strain energy density along the edges composing the elements, while force directions are along the unit vectors corresponding to the edges. Using this idea in subsequent work, authors [KRS16] introduce Graph-based Finite Element Analysis (Gra-FEA) where a damage variable corresponding to the edges of composing elements is used. It is based on the strain threshold and thus can simulate the fracture of an object by weakening the material. The advantage of using Gra-FEA is that it requires no remeshing, while adding little computational overhead on FEM and it still retains all the advantages of FEM. The dependency of Gra-FEA on the underlying mesh structure is thoroughly studied in another work [KRS19]. Our work introduces Gra-FEA to the visual simulation of fracture for computer graphics. Existing literature referred to above on Gra-FEA only focuses on 2D materials consisting of triangular elements with linear strain energy density. We extend the idea to the 3D domain and our fracture algorithm can handle linear as well as non-linear strain energy density. The original Gra-FEA formulation is restricted to simulating fracture for brittle materials. Our algorithm is capable of simulating fracture for both brittle and ductile materials. We believe, to the best of our knowledge, that this is the first work in computer graphics to simulate the fracture of 3D brittle, as well as ductile objects, with linear or non-linear strain energy density using graph-based FEM.

3. Governing Methods

We derive the formulation of graph-based FEM from the theory of classical FEM, in brief. Then we delve into the details of fracture generation using graph-based FEM. In our work, we always use tetrahedral finite elements with a linear basis for domain discretization.

3.1. Classical FEM

Consider a three-dimensional domain $\Omega \in \mathbb{R}^3$ that is discretized into a mesh of n_{tet} tetrahedra. Moreover, let \mathcal{N} be the set of nodes n_v shared by the tetrahedral mesh elements Δ_e . A displacement

Symbol	Definition
$\mathbf{F} = \mathbf{I} + \nabla_{\boldsymbol{\xi}} \mathbf{u}$	Deformation gradient
$J = \det(\mathbf{F})$	Relative volume change
$\mathbf{C} = \mathbf{F}^T \mathbf{F}$	Right Cauchy-Green deformation tensor
$I_C = \operatorname{trace}(\mathbf{C})$	First Right Cauchy-Green invariant
$II_C = \mathbf{C} : \mathbf{C}$	Second Right Cauchy-Green invariant
$III_C = \det(\mathbf{C})$	Third Right Cauchy-Green invariant
$\mathbf{E} = \frac{1}{2} \left(\mathbf{F}^T \mathbf{F} - \mathbf{I} \right)$	Lagrangian finite strain tensor
$\mathbf{P}(\mathbf{F}) = \frac{\partial \Psi}{\partial \mathbf{F}}$	First Piola-Kirchhoff stress tensor

Table 1: Quantities frequently used in FEM

function, $\mathbf{u} : \Omega \times [0, \infty) \longrightarrow \mathbb{R}^3$, can be defined as a mapping from a material point $\xi \in \Omega$ to its deformed location $\mathbf{x} \in \Omega_t$ in the world space. Ω_t represents the world space at time *t*. At a certain timestep $t \in [0, \infty)$, the displacement function can be represented as

$$\mathbf{u}(\boldsymbol{\xi},t) = \sum_{i \in \mathcal{N}} \mathbf{N}_i(\boldsymbol{\xi}) \mathbf{u}_i(t), \tag{1}$$

where $N_i(\xi)$ is the shape function and u_i is the displacement vector at the node *i*. For the sake of brevity of notation, we drop the shape (ξ) and time (*t*) parameters for the shape and displacement functions in further discussion.

The core idea of deformable object simulation is that when an object mesh is deformed from its rest position, a hyper-elastic strain energy density Ψ , develops inside it. The strain energy density produces internal elastic force \mathbf{f}^{int} , which in turn tries to restore the rest shape of the mesh. Thus, the system dynamics of a deformed object can be written in Lagrange's form as

$$\mathbf{M}\ddot{\mathbf{u}} + \mathbf{H}\dot{\mathbf{u}} + \mathbf{f}^{int} = \mathbf{f}^{ext} \tag{2}$$

$$\overline{\mathbf{u}} = \left(\mathbf{u}_1^T \dots \mathbf{u}_{n_v}^T\right)^T \tag{3}$$

Here **M** and **H** are the mass and damping matrices of the full system. The external and internal force vectors are represented by \mathbf{f}^{ext} and \mathbf{f}^{int} respectively.

Interested readers can look into the review papers [SB12] [KE20] for further details. In this work, we developed our fracture simulation model based on works by Smith et al. [SGK18], Sin et al. [SSB13] and Bargteil et al. [BWHT07]. Table 1 summarizes some of the quantities that we use frequently in FEM simulations.

3.2. Graph-based FEM

Let the hyper-elastic strain energy density of a tetrahedral element, Δ_e , be Ψ^e and $\mathbf{f}_{e_i}^{int}$ be the internal elastic force acting on the i^{th} node of the element. Now it can be proved [RS15] that nodal internal elastic forces of Δ_e can always be decomposed along the edge vectors connecting the nodes of Δ_e . The statement can be mathe-



Figure 3: Fracture of an armadillo model made of solid blue jade (top row) and porcelain shell (bottom row).



Figure 4: Fracture of a bunny model made of solid glass (top row) and a porcelain shell (bottom row).

matically formulated as

$$\mathbf{f}_{e_i}^{int} = V_e \frac{\partial \Psi^e}{\partial \mathbf{r}_i^e} = 2V_e \sum_{\substack{j=1\\i\neq i}}^{n_e} \frac{\partial \Psi^e}{\partial \left(d_{ij}^e\right)^2} d_{ij}^e \hat{\mathbf{d}}_{ij}^e \tag{4}$$

where \mathbf{r}_i^e is the world position vector for node *i* of element Δ_e . The parameter $d_{ij}^e = ||\mathbf{r}_i^e - \mathbf{r}_j^e||$ denotes the distance between the *i*th and *j*th nodes of Δ_e and $\hat{\mathbf{d}}_{ij}^e$ is unit vector along d_{ij}^e . Moreover, V_e and n_e represent the volume and the number of nodes of Δ_e respectively. In Equation 4 we can get to the rightmost term from the middle term by a few algebraic manipulations. We first take derivatives with respect to squared edge lengths and then apply a chain rule. Please check the supplemental document for a more detailed derivation. It can be seen from Equation 4 that the magnitude of the nodal internal elastic forces depends only on the derivative of the strain energy density of the element while the force directions are along its edges.

Figure 5 shows two tetrahedral elements, $\Delta_e ABDC$ and $\Delta_e BEDC$, which share a common face. These can be thought of as being a part of a larger object mesh. We assume that the overall mesh is deformed and each node in the mesh develops internal elastic force. Now according to Equation 4, internal elastic force on the



Figure 5: Nodal force distribution along the edges for graph-based FEM.

node *B* for $\Delta_e ABDC$ can be written along the edges of $\Delta_e ABDC$ as $\mathbf{f}_{\Delta_e ABDC}^B = \mathbf{f}_{\Delta_e ABDC}^{BC} + \mathbf{f}_{\Delta_e ABDC}^{BA} + \mathbf{f}_{\Delta_e ABDC}^{BD}$ (shown as dotted blue lines). Similarly, internal elastic force on the node *B* for $\Delta_e BEDC$ can be written along the edges of $\Delta_e BEDC$ as $\mathbf{f}_{\Delta_e BEDC}^B = \mathbf{f}_{\Delta_e BEDC}^{BC} + \mathbf{f}_{\Delta_e BEDC}^{BD} + \mathbf{f}_{\Delta_e BEDC}^{BE}$ (shown as dotted green lines). The distribution of nodal internal elastic forces for the entire mesh can be computed similarly. However, for Equation 4 to hold, we need to show that hyperelastic strain energy density Ψ^e can be completely represented using just d_{ij}^e . We prove this in the next section.

3.3. Proof for Edge Length Dependence of Strain Energy Density

Along with the Right Cauchy-Green deformation tensor **C** and the three invariants I_C , II_C and III_C , as defined in Table 1, let us introduce two more well-known anisotropic invariants [WMG96]

$$IV_C = \mathbf{a}^T \mathbf{C} \mathbf{b}$$

$$V_C = \mathbf{a}^T \mathbf{C}^T \mathbf{C} \mathbf{b}$$
(5)

where **a**, **b** are constant anisotropic fiber directions and they may or may not be equal to each other. Hyper-elastic energy density can generally be represented using a subset of these five invariants as $\Psi^e = f(I_C, II_C, III_C, IV_C, V_C)$. Therefore, in order to show that the hyper-elastic strain energy density can be expressed as a function of the length of edges of a mesh, it is sufficient to show that every element of the set of invariants can be expressed in closed form using only the length of the edges of the mesh in a graph-based FEM setting.

Theorem 3.1. Every element of the set of invariants $I_V = \{I_C, II_C, II_C, V_C\}$ can be expressed in closed form using only the length of the edges of a mesh used in FEM.

We present the complete proof of the theorem in the supplemental document.

3.4. Fracture with Graph-based FEM

In this section, we present our model for edge-based fracture derived from Graph-based FEM. We take advantage of the fact that the stress of an element is always a function of hyper-elastic strain energy density. We can find normal stress in the direction of the edges of a tetrahedral element by using a suitable transformation. Next, depending on the state of the edge i.e., broken or unbroken, we manipulate the normal stress along the edges to degrade the strain energy density for a damaged element. Finally, using this updated strain energy density, we recalculate the elastic forces and tangent stiffness matrix for fracture simulation.

Using chain rule of differentiation and deformation gradient from Table 1, we can rewrite the Equation 4 as

$$\mathbf{f}_{e_i}^{int} = 2V_e \sum_{\substack{j=1\\j\neq i}}^{n_e} \sum_{l=1}^{3} \sum_{l=1}^{3} \left[\frac{\partial \Psi^e}{\partial [\mathbf{F}]_{kl}} \frac{\partial [\mathbf{F}]_{kl}}{\partial \left(d_{ij}^e\right)^2} \right] d_{ij}^e \hat{\mathbf{d}}_{ij}^e \tag{6}$$

where $\frac{\partial \Psi^e}{\partial [\mathbf{F}]_{kl}}$ are components of first Piola-Kirchhoff stress tensor

of Δ_e (see Table 1). Thus, the edge-based elastic forces can be completely represented as a function of stress in Δ_e . This stress in turn depends on the derivative of element strain energy density, Ψ^e w.r.t. the deformation gradient **F**. Therefore, by manipulating the stress components properly we can change the internal elastic forces and in turn simulate the fracture of an element.

Let the rectangular Cartesian components of the Piola-Kirchhoff stress tensor, σ_c^e , be denoted as

$$\boldsymbol{\sigma}_{c}^{e} = \begin{bmatrix} \boldsymbol{\sigma}_{xx}^{e} & \boldsymbol{\sigma}_{yy}^{e} & \boldsymbol{\sigma}_{zz}^{e} & \boldsymbol{\sigma}_{xy}^{e} & \boldsymbol{\sigma}_{xz}^{e} & \boldsymbol{\sigma}_{yz}^{e} \end{bmatrix}$$
(7)

The set of normal stress along the edges is represented as

$$\mathbf{\sigma}_{d}^{e} = \begin{bmatrix} \mathbf{\sigma}_{12}^{e} & \mathbf{\sigma}_{13}^{e} & \mathbf{\sigma}_{14}^{e} & \mathbf{\sigma}_{23}^{e} & \mathbf{\sigma}_{24}^{e} & \mathbf{\sigma}_{34}^{e} \end{bmatrix}$$
(8)

where σ_{ij}^{e} represents the normal stress along the edge formed by nodes *i* and *j*, in Equation 8. Then the transformation of Cartesian stress to normal stress in the direction of an edge can be formulated as below [RS15] [KRS16]

$$\sigma_{ij}^{e} = \sigma_{xx}^{e} \cos^{2} \phi_{x} + \sigma_{yy}^{e} \cos^{2} \phi_{y} + \sigma_{zz}^{e} \cos^{2} \phi_{z} + \sigma_{xy}^{e} \cos \phi_{x} \cos \phi_{y} + \sigma_{xz}^{e} \cos \phi_{x} \cos \phi_{z} + \sigma_{yz}^{e} \cos \phi_{y} \cos \phi_{z}$$

$$\tag{9}$$

Assuming that $\hat{\mathbf{d}}_{ij}^{e}, \hat{\mathbf{x}}, \hat{\mathbf{y}}$ and $\hat{\mathbf{z}}$ denote unit vectors along the edge formed by nodes *i* and *j* of the tetrahedral element Δ_e , *x*-axis, *y*-axis and *z*-axis respectively, the following relations hold true

$$\cos\phi_x = \hat{\mathbf{d}}_{ij}^e \cdot \hat{\mathbf{x}}, \ \cos\phi_y = \hat{\mathbf{d}}_{ij}^e \cdot \hat{\mathbf{y}}, \ \cos\phi_z = \hat{\mathbf{d}}_{ij}^e \cdot \hat{\mathbf{z}}$$
(10)

Using Equation 9, we can write the final relation between two sets of stresses as

$$\begin{bmatrix} \sigma_{12}^{e} & \sigma_{13}^{e} & \sigma_{14}^{e} & \sigma_{23}^{e} & \sigma_{24}^{e} & \sigma_{34}^{e} \end{bmatrix}^{T} = \mathbf{T} \begin{bmatrix} \sigma_{xx}^{e} & \sigma_{yy}^{e} & \sigma_{zz}^{e} & \sigma_{xy}^{e} & \sigma_{xz}^{e} & \sigma_{yz}^{e} \end{bmatrix}^{T}$$
(11)

where, for convenience of notation, if we denote $\cos \phi$ as γ_{ϕ} , then

$$\mathbf{T} = \begin{bmatrix} \gamma_{\phi_{2}x}^{2} & \gamma_{\phi_{1}y}^{2} & \gamma_{\phi_{2}z}^{2} & \gamma_{\phi_{1}x}^{2}\gamma_{\phi_{1}y}^{2} & \gamma_{\phi_{2}x}^{1}\gamma_{\phi_{2}y}^{2} & \gamma_{\phi_{2}y}^{2}\gamma_{\phi_{2}z}^{2} \\ \gamma_{\phi_{1}x}^{2} & \gamma_{\phi_{1}x}^{2} & \gamma_{\phi_{2}z}^{2} & \gamma_{\phi_{2}x}^{2}\gamma_{\phi_{1}y}^{2} & \gamma_{\phi_{2}x}^{1}\gamma_{\phi_{2}y}^{2} & \gamma_{\phi_{2}x}^{1}\gamma_{\phi_{2}y}^{2} \\ \gamma_{\phi_{1}x}^{2} & \gamma_{\phi_{1}y}^{2} & \gamma_{\phi_{2}z}^{2} & \gamma_{\phi_{2}x}^{2}\gamma_{\phi_{2}y}^{2} & \gamma_{\phi_{2}x}^{2}\gamma_{\phi_{2}y}^{2} & \gamma_{\phi_{2}x}^{2}\gamma_{\phi_{2}x}^{2} & \gamma_{\phi_{2}y}^{2}\gamma_{\phi_{2}z}^{2} \\ \gamma_{\phi_{2}x}^{2} & \gamma_{\phi_{2}y}^{2} & \gamma_{\phi_{2}z}^{2} & \gamma_{\phi_{2}x}^{2}\gamma_{\phi_{2}y}^{2} & \gamma_{\phi_{2}x}^{2}\gamma_{\phi_{2}x}^{2} & \gamma_{\phi_{2}x}^{2}\gamma_{\phi_{2}x}^{2} & \gamma_{\phi_{2}x}^{2}\gamma_{\phi_{2}x}^{2} \\ \gamma_{\phi_{2}x}^{2} & \gamma_{\phi_{2}y}^{2} & \gamma_{\phi_{2}x}^{2} & \gamma_{\phi_{2}x}^{2}\gamma_{\phi_{2}x}^{2} & \gamma_{\phi_{2}x}^{2}\gamma_{\phi_{2}x}^{2} & \gamma_{\phi_{2}x}^{2}\gamma_{\phi_{2}x}^{2} \\ \gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{2}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{2}x}^{2} \\ \gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{2}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{2}x}^{2} \\ \gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2} \\ \gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2} \\ \gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2} \\ \gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2} \\ \gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2} \\ \gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2} \\ \gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2} & \gamma_{\phi_{3}x}^{2}\gamma_{\phi_{3}x}^{2} \\ \gamma_{$$

The matrix \mathbf{T} is calculated at every timestep and it is invertible for non-degenerate tetrahedral elements, i.e., elements with non-zero volume.

The criteria for fracture in graph-based FEM is that if the weighted average of normal stress along the direction of any edge exceeds the critical stress threshold σ_{thres}^{e} , a crack is to be formed on that edge. First, we explain the method to calculate the weighted average of normal stress. Let \mathbf{r}_{cen} be the position of the centroid of any tetrahedron of interest, Δ_{cen} , in the mesh at a particular time step t' during simulation. The weighted average of normal stress for Δ_{cen} can then be calculated as

$$\sigma_{cen}^{e} = \sum_{||\mathbf{r} - \mathbf{r}_{cen}|| \le R_{d}} \omega(\mathbf{r} - \mathbf{r}_{cen}) \sigma_{c}^{e}(\mathbf{r})$$
(13)

where ω is a weight kernel centered at \mathbf{r}_{cen} and R_d is support of the kernel. In our current implementation, we use a simple linear average kernel for ω for easier calculation. Other kernels e.g., a Gaussian kernel, can also be used. The exact diffusion fracture pattern (explained in the next paragraph) will be different for different kernels. Vector \mathbf{r} and tensor $\sigma_c^e(\mathbf{r})$ respectively refer to the position of centroid and PK1 stress tensor of any tetrahedron residing inside the kernel support R_d , at the same time step t'.

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Figure 6: Effect of kernel size R_d on fracture. We show the tearing of a PVC doormat (top) with different sizes of kernel support ($R_d = 0$ in middle, $R_d = 5$ at the bottom). As the size of kernel support increases, the cracks become more diffused over the object mesh.

Using Equations 11 and 13 the fracture criteria can, therefore, be defined as [KRS19]

$$\boldsymbol{\sigma}_{ij}^{e^*} = \left[\mathbf{T} \left[\boldsymbol{\sigma}_{cen}^{e} \right]^T \right]_{ij} \ge \boldsymbol{\sigma}_{thres}^{e} \tag{14}$$

where $\sigma_{ij}^{e^-}$ is a component of the weighted average of normal stress along the edge formed by nodes *i* and *j*. As the support of kernel, R_d , increases, the cracks become more diffused i.e., they spread more inside the object [KRS19]. For $R_d = 0$, the fracture criterion depends only on the stress value of the tetrahedron of interest, Δ_{cen} , resembling a local fracture. The effect of kernel size R_d on fracture is shown in Figure 6. The top image shows an undamaged doormat model. In the middle and bottom images, the doormat has been torn apart by applying the same force at the free left end. As evident from the figure, in the middle image, where $R_d = 0$, the fracture is localized in a nearly horizontal tear at one place. On the contrary, in the image at the bottom, where $R_d = 5$, the cracks have been propagated over the entire body of the doormat, in a diffused pattern.



Figure 7: Fractured edges shared multiple elements.

Now for each edge of the mesh, we maintain a binary damage

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variable, $\zeta \in \{0, 1\}$. A value of $\zeta = 0$ corresponds to an unbroken edge, while $\zeta = 1$ implies a crack is formed on the edge.

$$\zeta = \begin{cases} 0 & \text{if } \sigma_{ij}^{e^*} < \sigma_{thres} \\ 1 & \text{if } \sigma_{ij}^{e^*} \ge \sigma_{thres} \end{cases}$$
(15)

The number of broken edges in a tetrahedral element may vary from zero to six. Once an edge is broken, it will not get repaired in subsequent iterations and all the tetrahedra that share the broken edge will be considered damaged. As an example, a shared edge fracture is depicted in Figure 7, using a simplified, two-dimensional example. Two edges in the mesh, coloured red, are broken. These broken edges are shared by triangles 1 (two edges), 2 (one edge) and 3 (one edge). In the subsequent iterations, all three triangular elements will be considered damaged with varying degrees of damage. Whenever an edge, e_{br} , gets broken, the corresponding normal stress of that edge becomes zero for all elements sharing that edge. Using Equations 11 and 15, we can write

$$\begin{bmatrix} \sigma_{12_{frac}}^{e} & \sigma_{13_{frac}}^{e} & \sigma_{14_{frac}}^{e} & \sigma_{23_{frac}}^{e} & \sigma_{24_{frac}}^{e} & \sigma_{34_{frac}}^{e} \end{bmatrix}^{T} \\ = \operatorname{diag} \begin{bmatrix} \zeta_{12} & \zeta_{13} & \zeta_{14} & \zeta_{23} & \zeta_{24} & \zeta_{34} \end{bmatrix} \qquad (16) \\ \begin{bmatrix} \sigma_{12}^{e} & \sigma_{13}^{e} & \sigma_{14}^{e} & \sigma_{23}^{e} & \sigma_{24}^{e} & \sigma_{34}^{e} \end{bmatrix}^{T} \quad \forall e_{br} \in \Delta_{e} \end{bmatrix}$$

where ζ_{ij} and σ_{ijfrac}^{e} are damage variable and stress after fracture along the edge formed by nodes *i* and *j*. In matrix form Equation 16 is written as

$$\left[\boldsymbol{\sigma}_{d_{frac}}^{e}\right]^{T} = \zeta \left[\boldsymbol{\sigma}_{d}^{e}\right]^{T} = \zeta \mathbf{T} \left[\boldsymbol{\sigma}_{c}^{e}\right]^{T}$$
(17)

where **T** is transformation matrix from Equation 12. Projecting back to Cartesian space, rectangular Cartesian components, $\sigma_{c_{frac}}^{e}$ of the damaged Piola-Kirchhoff stress tensor can be written as

$$\left[\boldsymbol{\sigma}_{c_{frac}}^{e}\right]^{T} = \mathbf{T}^{-1} \left[\boldsymbol{\sigma}_{d_{frac}}^{e}\right]^{T} = \mathbf{T}^{-1} \boldsymbol{\zeta} \mathbf{T} \left[\boldsymbol{\sigma}_{c}^{e}\right]^{T}$$
(18)

Graph-based FEM may seem similar to mass-spring systems and peridynamics, but it has a few key differences as explained in the supplemental document.

3.5. Fractured Strain Energy Density: Linear Elasticity

In graph-based FEM, instead of remeshing the initial mesh, we update the system equation of fractured mesh using a reformulation of elastic energy density to simulate fracture. In this section, we present the detailed derivation of hyper-elastic strain energy density for graph-based FEM in the case of linear elasticity problems. First, we derive it for undamaged condition. The expression for the damaged condition is a simple extension of that.

3.5.1. Undamaged Condition

In linear elasticity, the strain energy density for a tetrahedral element, Δ_e , the nodal internal force vector \mathbf{f}_e^{int} of volume V_e can be written in terms of edge lengths as

$$\mathbf{f}_{e}^{int} = V_{e}\boldsymbol{\sigma}_{c}^{e}\mathbf{A}_{2}^{T}\mathbf{A}_{1}^{T} = V_{e}\boldsymbol{\sigma}_{c}^{e}\mathbf{B} = V_{e}\boldsymbol{\varepsilon}_{c}^{e}\mathbf{E}\mathbf{B}$$
(19)

where $\mathbf{A}_1 = \left[\frac{\partial \mathbf{I}_d^e}{\partial \mathbf{u}_e}\right]^T$, $\mathbf{A}_2 = \left[\frac{\partial \boldsymbol{\varepsilon}_c^e}{\partial \mathbf{I}_d^e}\right]^T$ and $\mathbf{B}^T = \mathbf{A}_1 \mathbf{A}_2$. Linear stress, $\boldsymbol{\sigma}_c^e$ and linear Cartesian strain, $\boldsymbol{\varepsilon}_c^e$ are given by $\boldsymbol{\sigma}_c^e = \boldsymbol{\varepsilon}_c^e \mathbf{E}$ and $\boldsymbol{\varepsilon}_c^e = 1/2(\mathbf{F} + \mathbf{F}^T) - \mathbf{I}$. The parameter \mathbf{I}_d^e denotes the edge length vector.

Similarly, we can formulate the tangent stiffness matrix in the graph-based FEM paradigm as below.

$$\mathbf{k}_{e} = \frac{\partial \mathbf{f}_{e}^{int}}{\partial \mathbf{u}_{e}} = V_{e} \mathbf{A}_{1} \mathbf{A}_{2} \mathbf{E} \mathbf{A}_{2}^{T} \mathbf{A}_{1}^{T} = V_{e} \mathbf{B}^{T} \mathbf{E} \mathbf{B}$$
(20)

More detailed derivations are given in the supplemental document.

3.5.2. Damaged Condition

Following the same line of arguments as presented in Equation 19 and Equation 20, the internal force and tangent stiffness matrix for fractured elements can be represented as

$$\mathbf{f}_{e_{frac}}^{int} = V_e \boldsymbol{\varepsilon}_c^e \mathbf{T}^T \boldsymbol{\zeta} \mathbf{T}^{-T} \mathbf{E} \mathbf{T}^{-1} \boldsymbol{\zeta} \mathbf{T} \mathbf{B}$$
(21)

$$\mathbf{k}_{e_{frac}} = V_e \mathbf{B}^T \mathbf{T}^T \boldsymbol{\zeta} \mathbf{T}^{-T} \mathbf{E} \mathbf{T}^{-1} \boldsymbol{\zeta} \mathbf{T} \mathbf{B}$$
(22)

More details are available in the supplemental document.

3.6. Fractured Strain Energy Density: Non-linear Elasticity

Non-linear elasticity demands a different reformulation of the elastic energy density. In this section, we present two different approaches for this reformulation that work in the case of non-linear elasticity. There are also some approaches we tried that did not produce the correct result. These are instructive for understanding and are presented in the supplemental document for completeness.

3.6.1. First Approach — Linearization

Our first approach is based on the linearization of non-linear strain energy density. We reparameterize the material parameters μ and λ to reproduce the stress tensor, σ_c^e , of linear elasticity according to Hooke's law.

$$\sigma_c^e = 2\mu_{\text{Lamé}}\varepsilon_c^e + \lambda_{\text{Lamé}}\text{trace}\left(\varepsilon_c^e\right)\mathbf{I}$$
(23)

where coefficients $\mu_{\text{Lamé}}$ and $\lambda_{\text{Lamé}}$ are Lamé parameters in linear elasticity. The parameter $\varepsilon_c^e = 1/2(\mathbf{F} + \mathbf{F}^T) - \mathbf{I}$ is linear strain. Given Young modulus (Y) and Poisson's ratio (v), Lamé parameters can be defined as

$$\mu_{\text{Lamé}} = \frac{\mathbf{Y}}{2(1+\nu)}$$

$$\lambda_{\text{Lamé}} = \frac{\mathbf{Y}\nu}{(1+\nu)(1-2\nu)}$$
(24)

For Neo-Hookean energy density [SGK18] in Equation 25, if we set $\mu = 4/3\mu_{\text{Lamé}}$ and $\lambda = \lambda_{\text{Lamé}} + 5/6\mu_{\text{Lamé}}$, it becomes consistent [SGK18] with Equation 23. With these linearized stress and strain values, we can rewrite the linearized Neo-Hookean energy density expression and follow the same line of arguments as presented earlier for fracture simulation.

$$\Psi_{neo} = \frac{\mu}{2} (I_C - 3) + \frac{\lambda}{2} (J - \alpha)^2 - \frac{\mu}{2} \log (I_C + 1)$$
 (25)

where
$$\alpha = 1 + \frac{\mu}{\lambda} - \frac{\mu}{4\lambda}$$
.

Similarly for Saint–Venant Kirchhoff energy density given in Equation 26, linearization [KTY09] can be done by putting $\mu = \mu_{\text{Lamé}}$ and $\lambda = \lambda_{\text{Lamé}}$. We have simulated fracture using this method on Saint–Venant Kirchhoff energy density in Figure 8.

$$\Psi_{stvk} = \frac{\lambda}{8} \left(I_C - 3 \right)^2 + \frac{\mu}{4} \left(II_C - 2I_C + 3 \right)$$
(26)



Figure 8: Linearization of strain energy density for fracture simulation on a 2D bar (1st row). The corresponding strain profiles are shown in $2^{nd}(\sigma_x)$, $3^{rd}(\sigma_{xy})$ and $4^{th}(\sigma_y)$ row respectively.

The derivations for the linearization of these strain energies are given in the supplemental document.

The main advantage of using this formulation is that even if an element gets fractured into two or more disjoint fragments, the residual force continues working on the remaining intact edges. However, for linearization operation, we assume infinitesimal linearized strain tensor $\varepsilon = 1/2(\mathbf{F} + \mathbf{F}^T) - \mathbf{I}$ instead of Lagrangian finite strain tensor (see Table 1). As higher order terms of the strain tensor are ignored for linearization, it can not capture large deformations without introducing significant artefacts. Thus, compared to non-linear models which use Lagrangian finite strain tensor, it produces more deformation artefacts.

3.6.2. Second Approach — Monotonic Degradation

Our second approach is based on the monotonic degradation of non-linear strain energy density. It is common in fracture simulation to degrade the energy density by a monotonic degradation function to allow material separation [AMM09] [MSU15] [WFL*19]. Thus, using Equation 8 and Equation 16 the hyper-elastic strain energy density for fractured elements can be defined as

$$\Psi^{e}_{frac} = \chi^{e} \left(\sigma^{e}_{d_{frac}}, \sigma^{e}_{d} \right) \Psi^{e}$$
⁽²⁷⁾

The monotonic function $\chi^e \left(\sigma^e_{d_{frac}}, \sigma^e_d \right)$ signifies the fractured hyper-elastic strain energy density as a fraction of original one de-

pending on the number of damaged edges.

$$\chi^{e}\left(\sigma_{d_{frac}}^{e},\sigma_{d}^{e}\right) = \frac{\sum\limits_{\substack{i,j=1\\i< j}}^{4} |\sigma_{ij_{frac}}^{e}|}{\sum\limits_{\substack{i,j=1\\i< j}}^{4} |\sigma_{ij}^{e}|}$$
(28)

where *i* and *j* denote nodes of the tetrahedral mesh element, Δ_e . Further, this updated hyper-elastic strain energy density is used to calculate internal force and tangent stiffness matrix for fractured elements. We have simulated fracture using this method on Saint–Venant Kirchhoff energy density in Figure 9.



Figure 9: Degradation of hyper-elastic strain energy density for fracture simulation on a 2D bar (1st row). The corresponding strain profiles are shown in $2^{nd}(\sigma_x)$, $3^{rd}(\sigma_{xy})$ and $4^{th}(\sigma_y)$ row respectively.

The main advantage of this method is that we can use Lagrangian finite strain tensor for non-linear elasticity. Thus using this approach, large deformations can be better simulated without any artefacts contrary to the earlier linearization method. However, in this approach, we assign zero value to the fractured hyper-elastic strain energy density i.e. $\Psi_{frac}^e = 0$, in Equation 27 when an element fractures into two or more disjoint fragments. Thus, in this case, when an element produces two or more disjoint segments, no residual force remains in the intact edges. This is a drawback compared to the previous linearization model where residual force continues working on the remaining intact edges even for disjoint fragments. However, any edge in an object mesh is shared by multiple elements all of which may not be fully damaged. The internal forces from these undamaged elements continue to work on those intact edges and thus, keep the nodes of the undamaged edges together for a fully split tetrahedron.

From Figure 8 and Figure 9, we can see that there is no significant change in the visual output of the fracture simulation regardless of the reformulation method used. Moreover, irrespective of methods used for updating strain energy density after fracture, both of these approaches use the rich discrete differential geometric approach [Sri21] of graph-based FEM to simulate fracture. From the discussion of the two approaches, we can conclude that if our main objective is to simulate large deformation-based fracture, we should use the second approach. On the other hand, if our goal is to accurately capture the residual stress after fracture for some engineering applications, then the first approach should be preferred.

3.7. Plasticity

We follow the multiplicative plasticity model [BWHT07] for our work. In this model, the deformation gradient is split into an elastic and a plastic part

$$\mathbf{F} = \mathbf{F}_e \mathbf{F}_p \tag{29}$$

where forcing det $(\mathbf{F}_p) = 1$, the volume of the element is preserved. Starting from an initial identity matrix, \mathbf{F}_p is updated as follows

$$\mathbf{F}_{p} \longleftarrow \mathbf{F}_{p} \cdot \mathbf{V} \left(\det\left(\Sigma\right)^{-\frac{1}{3}} \Sigma \right)^{\beta} \mathbf{V}^{T}$$
(30)

where $\mathbf{U}\Sigma\mathbf{V}^{T}$ is SVD of \mathbf{F}_{e} . The exponent β is an function of current stress (σ^{e}), yield stress (σ^{e}_{thres}), flow rate (ν) and hardening parameter (δ, K)

$$\beta = \operatorname{clamp}\left\{\frac{\nu(||\sigma^e||_2 - \sigma^e_{thres} - K\delta)}{||\sigma^e||_2}, 0\dots 1\right\}$$
(31)

where v and *K* are user-defined parameters. The term $K\delta$ determines the work hardening or softening. The term δ is initialized with value zero and incremented in each time step by $\delta \leftarrow \delta + \Delta t ||\sigma^e||_2$.

4. Implementation

Next, We present details of how we create visualizations of the fracture simulation. Following that, we discuss collision detection for fractured pieces. Then we present a complete algorithm for our model of fracture simulation.



Figure 10: (a) Original mesh with a fracture, (b) Computational mesh (M_c), with damaged edges marked in blue, on which system dynamics get evaluated, (c) Mesh for visualization (M_v) is split and the fracture surface is reconstructed.

4.1. Surface Remeshing for Visualization

Our FEM computations are performed on the graph induced by the computational mesh, M_c , that is never remeshed. However, to visualize the fracture we need to split the mesh used for visualization. For that purpose, a separate visualization surface mesh, M_v , is maintained in addition to M_c and is the same as the outer surface of volumetric mesh initially. We clarify this distinction between the M_c and M_v meshes using a simple 2D example. Figure 10b depicts that when a fracture occurs, the system dynamics is

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Figure 11: Splitting of edge (left), face (middle) and tetrahedron (right) for visualization of fracture.

evaluated on \mathcal{M}_c with weakened elements. When we need to render the fractured elements, we split the \mathcal{M}_v mesh that is used for visualization, as shown in Figure 10c and reconstruct the fracture surface. Remeshing of \mathcal{M}_v does not affect \mathcal{M}_c . Moreover, the reconstruction of the fracture surface for visualization adds very little computational overhead to the overall simulation cost. We have already explained how the system dynamics is computed on \mathcal{M}_c in the presence of a fracture.

We now explain the details of how we split \mathcal{M}_{ν} for rendering. We use a node-associated four-split configuration of a tetrahedron, as shown in Figure 11c, for fracture visualization. When a tetrahedral element splits, the corresponding edges and faces of the element split too. We explore two different split strategies, as explained at the end of this section.

When normal stress along an edge, PE1PE2 (Figure 11a), crosses the critical threshold, it splits into two segments PEISEI and PE2SE2. PE1 and PE2 are original nodes in the mesh. Let us call them parent nodes. The other two newly generated nodes, SEI and SE2, are child nodes to PE1 and PE2, respectively. The child nodes do not contribute to the system dynamics but rather follow the movements of their parent nodes. Similarly, when a face gets fractured into three segments (Figure 11b), three child nodes, SF1, SF2 and SF3 are assigned to their corresponding parent nodes, PF1, PF2 and PF3. Similarly four child nodes, ST1, ST2, ST3 and ST4 (Figure 11c) are assigned to four parent nodes PT1, PT2, PT3 and PT4 for a damaged tetrahedron. It is important to note that a triangular face or tetrahedron may not split into all three or four segments during the simulation. For example, with reference to Figure 11b, if we assume that only both the segments containing node PF1 get damaged i.e., edge PF2PF3 remain undamaged, then in such a case, child node SF1 follows the movement of parent node PF1 but child node SF2 and child node SF3 follow the average movement of parent nodes PF2 and PF3. When similar scenarios, although more complex, arise for a tetrahedron, we apply the same principle. All possible cases of different kinds of fractures are taken into account in our simulation.

In this work, we split edges, faces and tetrahedra with respect to their centroids. Furthermore, we also examined how the visualization output gets affected if an edge is split at a different position instead of in the middle. For this purpose, we split the edge such that the lengths of the sections are in inverse proportion to the ratio of its nodal forces. In Figure 12, in the left column, we render fracture on the bread model where all edges are split in the middle. In the right column, the same simulation is repeated with exact parameters and the edges are split in inverse proportion to the ratio of their corresponding nodal forces. As evident from the figure, there is no significant difference in the simulation outputs. We observed that both simulations require equivalent computation time.

4.2. Collision Detection

After the fracture, disjoint vertices have no internal elastic force acting on them. But external body forces e.g., gravity or impulse force, continue to be applied on all the vertices whether disjoint or not. So we need no special treatment for collision detection. Continuous Collision Detection (CCD) [TMT10] [TMOT12] [WFS*20] is performed between \mathcal{M}_{v} and the collider mesh (see Figure 3, 4). For fractured pieces colliding with the floor, we use Discrete Collision Detection (DCD) [ESHD05] routine (see Figure 4). When the visualization mesh, \mathcal{M}_{ν} , collides with another object mesh, we first calculate the impulse forces for each of them using the principles of CCD and DCD. Now, as shown in Figure 11c, each fractured segment of a tetrahedron has exactly one parent node. For each fractured segment, we accumulate all the impulse collision forces corresponding to that particular segment and apply the accumulated force to the parent node associated with the segment. However, as discussed in the previous section, for a partially fractured tetrahedron, there will be disjoint segments which contain more than one parent node. In that case, the accumulated force is equally distributed among all the parent nodes present in the fractured segments. Thus even though the visualization mesh, M_{ν} , accurately detects the collisions, the effects of the collisions are handled by the computational mesh, \mathcal{M}_{c} . Currently, we do not handle self-collision between the fractured pieces.

4.3. Algorithm

The full algorithm of our method is presented in Algorithm 1. We implement our model using the open-source Vega FEM library [SSB13]. In the implementation of this algorithm, for solving the system dynamics we use an implicit backward Euler integrator with a conjugate gradient solver [SSB13]. All the simulations are performed on an Intel Core i7-9750H CPU with 12 threads at 2.60 GHz. For visualization, the simulation results are raytraced in Houdini. Apart from the models obtained from



Figure 12: The left column displays fracture where an edge is split in middle. The right column displays fracture where an edge is split in the ratio of the forces on its nodes. The fractured areas in each image are zoomed up in the inset for better visualization.

Algorithm 1: Remeshing-Free Graph-based Fracture					
Initialize FEM simulation;					
Let n_v be total no of vertices of \mathcal{M}_c ;					
Let $[x]_{n_v \times 1}$ be initial position vector;					
while <i>True</i> do					
for Each element in \mathcal{M}_c do					
Calculate stress along the edges σ_{ij}^e [as per Eq. 9];					
if $\sigma_{ii}^e > \sigma_{thres}$ then					
Fracture the edge [as per Eq. 14];					
Update strain energy density Ψ^e [as explained					
in Sec. 3.5 & Sec. 3.6];					
Update internal force \mathbf{f}_{e}^{int} ;					
Update stiffness matrix \mathbf{k}_e ;					
Remesh the \mathcal{M}_{v} for visualization [as explained					
in Sec. 4.1];					
end					
Resolve all collisions with \mathcal{M}_{v} [as explained in					
Sec. 4.2];					
Calculate impulse force due to collision;					
Add all external forces to the vertices of \mathcal{M}_c ;					
end					
Build full system $[M]_{n_v \times n_v} [v]_{n_v \times 1} = [f]_{n_v \times 1};$					
Solve for velocity vector $[v]_{n_v \times 1}$;					
for Each vertex in \mathcal{M}_c and \mathcal{M}_v do					
Update position vector by $[x]_{n_v \times 1} + = \Delta t \cdot [v]_{n_v \times 1}$;					
end					
end					

the Stanford 3D scanning repository, the other 3D models used in this project are obtained from *free3d.com*, *sketchfab.com*, *turbosquid.com*, *cgtrader.com* and *lifesciencedb.jp*. Open-source software TetWild [HZG^{*}18] and TetGen [Si15] are used to generate the volumetric simulation meshes.

5. Results

We present multiple visualizations of fracture simulation for a variety of materials to demonstrate the robustness of our method and the diversity it can handle. We present dynamic fracture simulation results which mimic real-world events like tearing a loaf of bread, collision of hard objects with solid jade models and objects made of porcelain shells. We also show the effect of applying tensile or



Figure 13: Our method produces the intricate fracture patterns that result from the tearing of a loaf of bread. We show rendered frames from the simulation (to be seen left to right, top to bottom). The loaf model has around 620k tetrahedra.

impact forces to rubber-like and jello-like materials. Following that we discuss the effects of mesh resolution on both visualization and computational mesh during fracture simulation.

Next, we experimentally validate our model by recreating benchmark fracture experiments in simulation that are commonly performed in a structural mechanics laboratory, with our framework and compare the results with expected results from similar experiments performed in the real world. These experimental simulations are complete dynamics simulations that demonstrate the accuracy of our method. Finally, we compare our method with existing related works.

Mesh resolution and timing results for the simulations are reported in Table 2. The timestep for all our simulations is 5.0e - 03.

5.1. Dynamic Simulations

In a complete dynamic simulation, we take into account the effects of external forces and solve full system dynamics as presented in Equation 2 with mass & damping matrix regularizers. We use Neo-Hookean energy with monotonic degradation for ductile fracture simulation as it is better at preserving large deformation. StVK energy with linearization is used for brittle fracture simulation as brittle materials show very little deformation.

Simulation	# Tetrahedra	sec/frame	ρ (kg/m ³)	Y (Pa)	ν	R_d	σ_{th} (Pa)	$\sigma_p (Pa)$
Loaf (Fig. 1)	620.6k	4.38	1.0e+03	1.0e+07	0.35	1.0	4.0e+06	1.0e+06
PVC doormat (Fig. 6)	234.4k	1.62	1.0e+03	1.0e+08	0.45	0.0 & 5.0	2.5e+06	1.5e+05
Jello armadillo (Fig. 2)	159.1k	1.09	1.0e+03	1.0e+06	0.3	2.0	1.5e+05	5e+03
Solid jade armadillo (Fig. 3)	4.7k	0.04	1.0e+03	1.0e+10	0.4	0.0	5.0e+07	-
Porcelain shell armadillo (Fig. 3)	41.1k	0.33	1.0e+03	1.0e+10	0.4	0.0	5.0e+07	-
Solid glass bunny (Fig. 4)	23.7k	0.18	1.0e+03	1.0e+10	0.45	0.0	5.0e+07	-
Porcelain shell bunny (Fig. 4)	38.8k	0.29	1.0e+03	1.0e+10	0.45	0.0	5.0e+07	-
Granite cylinder (Fig. 21)	45.8k	0.39	2.75e+03	4.5e+10	0.33	0.0	9.56e+06	-
A516 Gr.70 steel bar (Fig. 17, 19)								
Brittle at temperature $-40^{\circ}C$	27.8k	0.24	7.7e+03	2.2e+11	0.28	0.0	6.55e+08	-
NVE 36 steel bar (Fig. 17, 19)								
Ductile at temperature $23^{\circ}C$	27.8k	0.24	7.7e+03	1.9e+11	0.28	0.0	5.4e+08	8.3e+07

Table 2: Simulation parameter table. Parameters ρ , **Y**, ν and R_d (Equation 13) denote density, Young modulus, Poisson's ratio and support of the kernel for crack diffusion. Parameters σ_{th} and σ_p denote yield and plasticity threshold.

5.1.1. Ductile Fracture

We illustrate the highly detailed and intricate fracture effects produced by tearing a loaf of bread in Figure 1 and Figure 13. It is evident from these figures that our simulation model can model complex fracture patterns and can easily scale to high resolution meshes. In Figure 2 we simulate the limbs of a jello armadillo being ripped off. Here we also visualize the corresponding strain profiles. In the frame shown on the leftmost side, the armadillo appears just prior to the fracture. Thus, it has the highest strain on elements that are going to be damaged (coloured red). Once damage sets in post-fracture, as shown on the right, the strain decreases again. The disconnected nodes in the damaged mesh have very low strain values and thus appear in blue-green. In Figure 6, we show the tearing of a doormat with different sizes of kernel support, R_d , thus producing localized (middle) and diffused (bottom) fracture.

5.1.2. Brittle Fracture

For brittle fracture simulation, over our regular graph-based FEM, we have used a stress-based fracture surface initiation and propagation algorithm similar to [KLB15] [GMD13]. Similar to those works, first we identify a few regions containing the highest amount of stress and then initiate a single fracture surface from each of these regions. Each of these fracture surfaces propagates to the boundary of the object. We first calculate the intersections of a fracture surface with the edges of our FEM mesh and then mark these edges as damaged with the embedded intersection points. Finally, the graph-based FEM is used to simulate the fractured object.

In the top row of Figure 3, a hammer hits an armadillo made of solid blue jade to generate large chunks of debris from the fracture. In the bottom row of the same figure, a hammer smashes a porcelain shell model of an armadillo.

Using our fracture simulation method, in the top row of Figure 4 we render the fracture of a solid glass model of a bunny when it is hit by a hammer. The fracture of a shell model of a bunny made of porcelain is simulated in the bottom row of the same figure.

5.2. Effect of Mesh Resolution

We show the effect of mesh resolution on the fracture simulation in Figure 14. In the figure, the computational mesh resolution is increased from top to bottom with topmost, middle and bottom-most models consisting of 1.5k, 38k and 620k tetrahedra respectively. As evident from the figure, with the increase of mesh resolution, finer and more intricate details of the fracture can be captured. The simulation times required are 0.013 sec/frame, 0.31 sec/frame and 4.38 sec/frame respectively from lowest to highest resolution mesh.



Figure 14: Computational mesh resolution is increased from top to bottom.

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5.3. Debris control

We can control the amount of debris produced due to fracture by tuning different parameters. In Figure 15, a bread fracture are shown with varying amount of debris. The parameters that we have tuned in this example to produce less debris are plasticity and fracture threshold. The image on the left has plasticity and fracture threshold 2.5e + 06 Pa and 3.0e + 06 Pa respectively. On the contrary, the plasticity and fracture threshold for the image on right is 1.0e + 06 Pa and 4.0e + 06 Pa respectively.



Figure 15: Fracture simulation with varying amount of debris. The left column shows more debris while the right column shows less amount of debris.



Figure 16: Fracture simulation with increasing (left to right) amount of debris.

For brittle fracture, as discussed earlier, first we pick out a few regions containing the highest amount of stress and then initiate a single fracture surface from each of these regions. Now, if more number of high stress regions are picked out, the amount of debris also increases. In Figure 16, the amount of debris for the brittle fracture of a porcelain bunny shell is depicted. In the figure, we choose 50 high stress regions for the leftmost image, 150 high stress regions for the middle image and 300 high stress regions for the rightmost image. It can be seen in the figure that the amount of debris increases from left to right with the increasing number of high stress regions. The user can easily change this parameter to control debris.

5.4. Experimental Validation

To evaluate and validate how faithfully real-world fracture can be simulated using our method, we present a qualitative and quantitative comparison with three benchmark laboratory fracture experiments. These are the Charpy impact test, the Izod impact test and the Brazilian test.

5.4.1. Charpy Impact Test

In this test, we use a steel specimen of dimension $10mm \times 10mm \times 55mm$ with a 45° 'V' shaped groove in the middle of it (see the left image of Figure 17). A pendulum of known mass and length then impacts the back of the groove with both ends of the specimen held against fixed beams. Now depending on the tensile strength and plasticity threshold of the material, the specimen breaks fully



Figure 17: **Charpy Impact Test**: The left image shows the configuration of the Charpy test. In the middle image, a specimen with a higher plasticity threshold splits into two, while on the right a specimen with a low plasticity threshold only bends and gets partially damaged.



Figure 18: The plot shows the load-displacement curves of our simulated Charpy impact test. The simulated curves are compared with ground truth curves from actual laboratory experiments.

into two pieces or is partially damaged and bends in a three-point configuration. We simulate this experiment using our method, on specimens made of two different kinds of steel with different tensile strengths (the name of the steel is mentioned in Table 2). Instead of the entire swinging pendulum, as present in the real Charpy test setup, we simulate an impact with a fast-moving block (as can be seen in the image) that hits the specimen at the same location where the specimen is supposed to be hit in the real test. As shown in Figure 17 the steel with a higher plasticity threshold (middle image) breaks into two pieces while the other one (right image) splits partially and bends. Simulation material parameters used in this experiment are obtained from [TML02].

5.4.2. Izod Impact Test

The Izod impact test specimen is shown in the left image of Figure 19. Here the specimen is held in a cantilever beam configuration with one end fixed and the pendulum hits it on the other end. The results of the Izod test performed on the same materials as before are shown in the middle and right images of the figure.

In Figure 18 and Figure 20 we plot the load-displacement curves for the Charpy and Izod tests respectively as obtained from our simulated experiments. The solid blue curve denotes simulated brittle fracture, which splits sharply beyond a load threshold. The



Figure 19: **Izod Impact Test**: The left image shows the configuration of the Izod test. In the middle image, a specimen with a higher plasticity threshold splits into two, while on the right a specimen with a low plasticity threshold only bends and gets partially damaged.



Figure 20: The plot shows the load-displacement curves of our simulated Izod impact test. The simulated curves are compared with ground truth curves from actual laboratory experiments.

solid red curve which denotes simulated ductile fracture indicates a plastic flow in the material. The brittle and ductile fracture curves for both tests closely resemble the ground truth reference forcedisplacement curves shown in the same figure with dotted lines. The ground truth reference curves are reproduced from the same experiments reported in [TML02]. The reference data for ductile fracture is obtained from Figure 10 in [TML02]. In this figure, both the force/load and load-line displacement are plotted against the time axis. However, in our plots, we depicted the force/load curve against the load-line displacement. Because in standard practice, the ductility and brittleness of material are denoted using the loaddisplacement curve as illustrated in Figure 4 in [TML02]. Similarly, the reference data for brittle fracture is obtained from Figure 12 in [TML02]. Brittle material fractures completely after the elastic region whereas ductile material partially fractures and then bends due to plasticity. This proves that brittle and ductile material simulation performed by our method closely matches the behaviour of real-world materials, both qualitatively and quantitatively.

5.4.3. Brazilian Test

The Brazilian test is a laboratory test for the indirect measurement of tensile strength. In this test, a vertical load is applied at the highest point of a cylinder, the axis of which is placed horizontally and is supported on a horizontal plane. Here we use a cylinder made of



Figure 21: **Brazilian test**: Initial (left) and fractured (middle & right) configuration of cement cylinder.



Figure 22: **Brazilian test**: Load displacement curve for original and simulated experiments.

granite. In Figure 21 the initial (left) and fractured (middle & right) configurations of the cylinder are shown. We compare the precision of the experiment performed in simulation using our method with a similar real experiment from literature [GBA*21] [Exp13] by plotting their corresponding load-displacement curves in Figure 22. The close match between the two curves validates the accuracy of our simulation model. Simulation material parameters used in this experiment are obtained from [GBA*21].

5.5. Comparison with Existing Techniques

To the best of our knowledge, this is the first work in computer graphics that presents a FEM-based fracture method that is graphbased, remeshing-free, highly stable and can incorporate a large number of cracks with little extra computational overhead on FEM.



Figure 23: **XFEM vs Our Method**: A cylinder is split with a single cut. This is simulated using XFEM (top row) and Graph-based FEM (bottom row). Graph-based FEM produces comparable results with significantly lower run-time.

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Figure 24: **XFEM vs Our Method**: A cylinder is split into many parts with multiple random cuts. This is simulated using XFEM (top row) and Graph-based FEM (bottom row).

We have compared our method with other existing fracture simulation algorithms in the literature.

5.5.1. XFEM vs Our method

Unlike XFEM [KBT17] [CMSK20] or VNA [SDF07] [WJST15], the size of the system matrix remains constant throughout the simulation, thus reducing the computational cost over standard FEM substantially. We have implemented the method presented by Koschier et al. [KBT17] and then compared it with our method. The visual comparison between XFEM and graph-based FEM is presented in Figure 23. As shown in the figure, we use a cylindrical object consisting of 600 DOFs (nodes) and 1566 tetrahedra with a pre-defined vertical split in the middle. To simulate the split, we need to enrich appropriate nodes for XFEM and relabel the appropriate edges as damaged for our graph-based FEM simulation. After enrichment, the number of DOFs (original nodes and enriched nodes together) for XFEM simulation increases to 852 while the number of DOFs remains the same in graph-based FEM. Next, we let the disjoint piece fall under gravity and calculate the average frame rate for both simulations. The simulation of XFEM takes around 0.271 sec/frame while our graph-based FEM requires 0.009 sec/frame. Similarly, in Figure 24, we use another cylindrical object consisting of 7.7k DOFs and 33.5k tetrahedra with multiple random splits. After enrichment, the number of DOFs for XFEM simulation increases to 9.2k while the number of DOFs remains the same in graph-based FEM. The simulation of XFEM takes around 5.87 sec/frame while our graph-based FEM requires 0.253 sec/frame.

We also present a numerical comparison between our method and XFEM-based fracture simulation by Chitalu et al. [CMSK20]. Using their open source code-base [Chi20], the XFEM-based method requires 2743 sec to render 29 number of cuts for a solid bunny model consisting of 23.7k tetrahedra. On the other hand, our graph-based FEM model requires only 0.18 sec to render the same number of cuts on the same bunny model. Thus, we have a clear advantage in terms of the speed of simulation. Moreover, simulating a large amount of debris with existing FEM or XFEM-based solutions in literature requires prohibitively expensive numerical computation due to remeshing and linear scaling of the system matrix. Our method can simulate a very high amount of fracture debris easily. Thus, our method enjoys the good characteristics of both FEM and XFEM methods, without the overheads.

Moreover, while rendering multiple random cuts using XFEM, we have noticed that quite often it generates skewed volume ratios of the tetrahedral elements. This in turn makes the system matrix degenerate with high condition number. Thus for a larger time step



Figure 25: **Stability of XFEM vs Our Method**: Multiple random splits of a cylindrical object are simulated using XFEM (top row) and Graph-based FEM (bottom row). For a larger time step $(\Delta t = 5 \times 10^{-2} \text{sec})$ XFEM simulation becomes unstable while our method remains stable.

 $(\Delta t = 5 \times 10^{-2} \text{sec})$ XFEM quickly becomes unstable (see Figure 25). On the contrary, as the system matrix remains unchanged throughout the simulation for graph-based FEM, our method can remain stable even for a considerably larger time step.

5.5.2. Remeshing-based FEM vs Our method

As no open source implementation of the work [OH99] that simulates fracture using FEM followed by appropriate remeshing, is available, we compare our method to it by reproducing the fracture based on the numerical data as reported in the paper. Brittle fracture of an adobe wall, which contains 338 nodes and 1109 elements initially, is rendered in [OH99]. After the fracture, the mesh size grows to 6892 nodes and 8275 elements in the final configuration with multiple disjoint segments. Plugging in the same set of parameters as reported in [OH99], Vega FEM takes 0.11 - 0.14sec/frame to simulate a similar number of disjoint objects, which have an equal number of nodes and elements together. Compared to this, our method can simulate the fracture of an object to produce the similar number of disjoint pieces in just 0.007 sec. Our object mesh consists of approximately 400 nodes and 1200 elements with the same set of material parameters as before. In our graph-based FEM, the number of nodes and tetrahedra remain unchanged even after fracture, resulting in a speed-up in computation time compared to remeshing-based methods. This shows a reduction by nearly 15 to 20 times in required simulation time using our method.

6. Discussions and Future Work

We present a novel graph-based remeshing-free FEM approach for ductile and brittle fracture. We derive theoretical proof to show that our method extends to non-linear hyper-elastic strain energy densities. We follow this with the complete algorithmic description of our model. The high stability, speed and robustness of our method are illustrated and established via multiple dynamic experiments with different materials. We evaluate the appeal and realism of our fracture simulations through results on objects made of a wide variety of materials. Comparing with benchmark fracture experiments, we validate the correctness and accuracy of our method. We also compare with existing XFEM simulation methods, providing quantitative and visual evidence about the better performance of our approach.

Even though we believe that our method is capable of producing particularly realistic fracture of a variety of materials at a high speed, it does have some limitations in its current form. Currently, an edge of an element in the object mesh can split into a maximum of two parts, thus limiting the fracture of tetrahedral elements to a maximum of four parts. We wish to extend our work to incorporate a high number of fracture segments in a single element. Moreover, even though our surface remeshing algorithm uses floating point operations for element splitting, we do not employ any safeguard to resolve rare errors that might occur due to floating point operations. Despite this, our algorithm is very stable, even when simulating highly complex fracture patterns. Some possible solutions could be employing the methods from [WJST15] [WFS*20] where authors resolve floating point errors by providing an appropriate error bound or by looking for increasingly tighter sub-intervals. We would like to model anisotropic fractures in future work.

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